

2. A flat plate of length L is aligned with a free stream flow $U\mathbf{i}$. Starting from the Navier-Stokes equations in the form

$$\nabla \cdot \mathbf{u} = 0, \quad (\mathbf{u} \cdot \nabla) \mathbf{u} = -\frac{1}{\rho} \nabla p + \nu \nabla^2 \mathbf{u},$$

show that if $UL/\nu \gg 1$, the flow near the plate can be described by the boundary layer equations

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = \frac{\partial^2 u}{\partial y^2}, \quad \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0, \quad (2.1)$$

when written in suitable non-dimensional variables. What boundary conditions have to be imposed? *Verify* that if the plate is semi-infinite, with the leading edge at the origin, then $u = f'(\eta)$, $v = -x^{-1/2}(f(\eta) - \eta f'(\eta))/2$, with $\eta = yx^{-1/2}$ satisfy (2.1) as long as

$$f''' + \frac{1}{2} f f'' = 0,$$

with $f(0) = f'(0) = 0$ and $f' \rightarrow 1$ as $\eta \rightarrow \infty$. Explain briefly why (2.1) should reduce to an ordinary differential equation in this case.

What extra term would appear in (2.1) if the plate is replaced by a finite blunt body? Give a rough sketch of the streamlines and compare them with those predicted by an inviscid irrotational model.

B6-a. 1998

Fluid Dynamics

1. The Navier-Stokes equations describing two-dimensional incompressible flow can be written in the form

$$\rho \left[\frac{\partial(\nabla^2 \psi)}{\partial t} + \frac{\partial(\psi, \nabla^2 \psi)}{\partial(y, x)} \right] = \mu \nabla^4 \psi,$$

where ψ is the stream function. When the characteristic velocity and length are U and L respectively, show that the dimensionless stream function satisfies

$$\frac{\partial(\nabla^2 \psi)}{\partial t} + \frac{\partial(\psi, \nabla^2 \psi)}{\partial(y, x)} = \frac{1}{Re} \nabla^4 \psi, \quad (*)$$

where $Re = \frac{\rho U L}{\mu}$ and ψ, x, y and t are now dimensionless.

Fluid flows steadily in a two-dimensional wedge defined by $-\alpha < \theta < \alpha$ in polar coordinates under the action of a source at the origin whose volume flow rate (per unit width transverse to the flow) is M . Show that a suitable scaling for the velocity at distance L from the source is $M/2\alpha L$ and that $Re = M\rho/2\alpha\mu$. Show that a possible flow is $\psi = f(\theta)$ where

$$f'''' + 4f'' + 2Re f' f'' = 0$$

with $f'(\pm\alpha) = 0$, $f(\alpha) - f(-\alpha) = 2\alpha$.

Find an approximate solution for $v = f'(\theta)$ when $Re \ll 1$, and sketch the graph of $v(\theta)$. How do you expect the profile for v to be modified if $Re \gg 1$?

[In plane polar coordinates $\nabla^2 = \frac{\partial^2}{\partial r^2} + \frac{1}{r} \frac{\partial}{\partial r} + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2}$.]

B.6.a 1999

2. Incompressible fluid of kinematic viscosity ν flows past a plane wall $y = 0$ with velocity $U(x) = U_0(x/L)^m$ where U_0 , L and m are constants. The stream function is ψ , which vanishes on the wall. State the conditions under which there is a boundary layer in which the dimensionless stream function Ψ satisfies

$$\frac{\partial \Psi}{\partial Y} \frac{\partial^2 \Psi}{\partial X \partial Y} - \frac{\partial \Psi}{\partial X} \frac{\partial^2 \Psi}{\partial Y^2} = \frac{\partial^3 \Psi}{\partial Y^3} + mX^{2m-1},$$

with

$$\frac{\partial \Psi}{\partial Y} \rightarrow X^m \quad \text{as } Y \rightarrow \infty, \quad \Psi = \frac{\partial \Psi}{\partial Y} = 0 \quad \text{on } Y = 0.$$

Also, define X, Y and Ψ in terms of x, y and ψ . Show that there is a similarity solution in which

$$\Psi = X^{(m+1)/2} f(Y/X^{(1-m)/2})$$

where

$$f''' + \frac{m+1}{2} f f'' = m(f'^2 - 1)$$

with $f(0) = f'(0) = 0$, $f'(\infty) = 1$. How does the fact that this problem only has stable solutions for $m > -0.09 \dots$ relate to the theory of flight?

B6.a 2000

Viscous Flow

1. A two-dimensional flow impinges on a fixed wall, $y = 0$, so that the inviscid approximation for the velocity near the stagnation point is $(\alpha x, -\alpha y)$. Show that if the length scale L in the x -direction is chosen such that $L \gg \sqrt{\frac{\nu}{\alpha}}$ where ν is the kinematic viscosity, there will be a boundary layer on the wall of width $O(\sqrt{\frac{\nu}{\alpha}})$ where the nondimensional equations for the stream function may be written as

$$\frac{\partial \psi}{\partial Y} \frac{\partial^2 \psi}{\partial X \partial Y} - \frac{\partial \psi}{\partial X} \frac{\partial^2 \psi}{\partial Y^2} = X + \frac{\partial^3 \psi}{\partial Y^3}$$

where X, Y are suitably scaled coordinates.

Write down the appropriate boundary conditions for ψ .

By substituting $\psi = X^\gamma f(Y/X^\beta)$ into this equation, show that such a similarity solution only exists if $\gamma = 1$, $\beta = 0$ and f satisfies

$$f''' + f f'' = f'^2 - 1,$$

with $f(0) = f'(0) = 0$ and $f'(\infty) = 1$.

B.6 a 2001

2. A uniform, incompressible stream of fluid of kinematic viscosity ν and speed U flows past a flat plate of length L . Define a Reynolds number and write down a suitable dimensionless form of the Navier–Stokes equations. When the Reynolds number is large, derive the two-dimensional boundary layer equations

$$\begin{aligned}u \frac{\partial u}{\partial x} + V \frac{\partial u}{\partial Y} &= \frac{\partial^2 u}{\partial Y^2}, \\ \frac{\partial u}{\partial x} + \frac{\partial V}{\partial Y} &= 0,\end{aligned}\tag{1}$$

where the non-dimensional velocities (u, V) and coordinates (x, Y) should be defined.

What boundary conditions have to be imposed?

If ψ denotes the stream function and the leading edge of the plate is at $x = 0$, show that a solution exists of the form $\psi = \sqrt{x} f(\eta)$, where $\eta = Y/\sqrt{x}$, as long as

$$f''' + \frac{1}{2} f f'' = 0,\tag{2}$$

with $f(0) = f'(0) = 0$ and $f' \rightarrow 1$ as $\eta \rightarrow \infty$.

Describe how the invariance of (2) under the transformation $f(\eta) \rightarrow af(a\eta)$ enables the boundary value problem for f to be recast as an initial value problem.

B6 a 2002.

2. Explain the physical significance of the terms in the energy equation

$$\rho c \frac{dT}{dt} = k \nabla^2 T + \Phi,$$

in the usual notation, where T is the fluid temperature. Does Φ increase or decrease with the viscosity and the strain-rate respectively? Show that in an inviscid uniform flow in the x -direction, the non-dimensional version of this equation is

$$\frac{\partial T}{\partial x} = \epsilon \nabla^2 T,$$

where ϵ is a constant.

Suppose that the ambient temperature is zero and that a plate at temperature $T = 1$ lies along $y = 0$, $x > 0$. Show that as $\epsilon \rightarrow 0$, there is a thermal boundary layer on the plate in which $Y = y/\sqrt{\epsilon} = O(1)$ and, to lowest order, T satisfies

$$\frac{\partial T}{\partial x} = \frac{\partial^2 T}{\partial Y^2}.$$

What boundary conditions must be satisfied by the solution to this equation? Show that

$$T = \frac{2}{\sqrt{\pi}} \int_{Y/2\sqrt{x}}^{\infty} e^{-s^2} ds$$

satisfies the equation and boundary conditions, and indicate on a sketch where you expect the solution to be valid.

Apply the same argument to show that if the dimensionless stream function ψ in a high Reynolds number flow past a plate lying along $y = 0$, $x > 0$ satisfies

$$\frac{\partial(\psi, \nabla^2 \psi)}{\partial(y, x)} = \epsilon \nabla^4 \psi$$

with $\frac{\partial \psi}{\partial y} \rightarrow 1$ as $x^2 + y^2 \rightarrow \infty$, then, in the boundary layer,

$$\left(\frac{\partial \Psi}{\partial Y} \frac{\partial}{\partial x} - \frac{\partial \Psi}{\partial x} \frac{\partial}{\partial Y} \right) \frac{\partial^2 \Psi}{\partial Y^2} = \frac{\partial^4 \Psi}{\partial Y^4}.$$

What is the relation between ψ and Ψ ? What are the boundary conditions on Ψ ? Where do you expect the boundary layer approximation to be valid?

B6.a 2003 .

2. The streamfunction in a two-dimensional flow past a flat plate $y = 0$ satisfies the dimensionless equation

$$\frac{\partial(\psi, \nabla^2 \psi)}{\partial(y, x)} = \frac{1}{\text{Re}} \nabla^4 \psi,$$

where Re is the Reynolds number. When $\text{Re} = \infty$, there is a slip velocity $U(x)$ on the plate. Show that, when Re is large but finite, the flow near the plate only differs appreciably from $U(x)$ in a boundary layer in which $Y = y\sqrt{\text{Re}} = O(1)$ and that, in this boundary layer,

$$\psi(x, y) = \frac{1}{\sqrt{\text{Re}}} \Psi(x, Y), \quad \text{where} \quad \frac{\partial(\Psi, \partial^2 \Psi / \partial Y^2)}{\partial(Y, x)} = \frac{\partial^4 \Psi}{\partial Y^4}.$$

As well as $\Psi(x, 0) = \partial \Psi / \partial Y(x, 0) = 0$, what other condition is needed to determine Ψ ?

Suppose the solution is a similarity solution in which $\Psi(x, Y) = f(x)g(Y/h(x))$. Show that $f = h^\lambda$, for some constant λ , and deduce that

$$h^\lambda \frac{dh}{dx} = \text{constant}.$$

Infer that either $U(x) \propto (x - x_0)^{(\lambda-1)/(\lambda+1)}$, for $\lambda \neq -1$, or $U(x) \propto e^{cx}$, where x_0 and c are constants. Deduce that the boundary layer thickness is only constant when $U(x)$ is linear in x . To what physical situation does this correspond?

36.a 2004

3. Suppose that a two-dimensional jet emanates in the x -direction from a nozzle at the origin into quiescent fluid. Starting from the steady dimensionless Navier-Stokes equations

$$(\mathbf{u} \cdot \nabla)\mathbf{u} = -\nabla p + \frac{1}{\text{Re}}\nabla^2\mathbf{u}, \quad \nabla \cdot \mathbf{u} = 0,$$

show that in $x > 0$ the jet is described by

$$\begin{aligned} u \frac{\partial u}{\partial x} + V \frac{\partial u}{\partial Y} &= -\frac{\partial p}{\partial x} + \frac{\partial^2 u}{\partial Y^2}, \\ 0 &= \frac{\partial p}{\partial y}, \\ \frac{\partial u}{\partial x} + \frac{\partial V}{\partial Y} &= 0, \end{aligned}$$

after a suitable rescaling. What are the appropriate boundary conditions on $Y = 0$ and as $Y \rightarrow \infty$? Show that the boundary layer equations imply

$$\int_{-\infty}^{\infty} u^2 dY = \text{constant}.$$

What is the physical interpretation of this result? What is the physical interpretation of the fact that $\int_{-\infty}^{\infty} u dY$ is not constant?

Show that there is a solution of the form $\Psi = x^{1/3}f(\eta)$ where $\eta = Y/x^{2/3}$ and Ψ is the streamfunction. What are the equation and boundary conditions satisfied by f ?

B6.a. 2005.

2. Prandtl's boundary layer equations are

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = \nu \frac{\partial^2 u}{\partial y^2} + U \frac{dU}{dx}, \quad \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0. \quad (1)$$

Define all the variables in these equations and state the conditions under which these equations hold.

Show that the complex potential $w = A \log z$ represents an inviscid flow in a wedge of angle α with a source of strength $A\alpha$ at the origin. Explain why, in high Reynolds number flow, boundary layers will develop on the fixed boundaries.

The boundaries of the flow are at $y = 0$ and $y = x \tan \alpha$. Show that there is a boundary layer on $y = 0$ where equations (1) hold with $U = Ax^{-1}$. What are the boundary conditions in this case? Show that there is a solution of the form $\psi = f(y/x)$ where ψ is the stream function and f satisfies

$$\nu f''' + (f')^2 = A^2$$

with $f(0) = f'(0) = 0$ and $f'(\eta) \rightarrow A$ as $\eta \rightarrow \infty$.

Show that this problem has no solution if $A > 0$. What are the implications of this result?

B6.a 2006

A. Viscous Flow

1. The non-dimensional form of the Navier-Stokes equation for steady flow is

$$(\mathbf{u} \cdot \nabla) \mathbf{u} = -\nabla p + \frac{1}{Re} \nabla^2 \mathbf{u},$$

$$\nabla \cdot \mathbf{u} = 0.$$

When the Reynolds number Re is large, derive Prandtl's boundary layer equations for a flow $\mathbf{u} = U(x)\mathbf{i}$ past a flat plate at $y = 0$, $x > 0$, in the form

$$u \frac{\partial u}{\partial x} + V \frac{\partial u}{\partial Y} = U \frac{dU}{dx} + \frac{\partial^2 u}{\partial Y^2}, \quad \frac{\partial u}{\partial x} + \frac{\partial V}{\partial Y} = 0,$$

where the variables Y , V should be defined.

When $U(x) = e^{\alpha x}$, show that there is a similarity solution for the stream function ψ of the form

$$\psi = e^{\beta x} f(Y e^{\gamma x}),$$

where β , γ should be determined, and f satisfies

$$\alpha f'^2 - \frac{\alpha}{2} f f'' = \alpha + f'''.$$

What are the boundary conditions for f ?

Deduce that the width of the boundary layer will increase/decrease as x increases according as the external flow is decelerating/accelerating in x . Is this observation true in more general flows? Is this boundary layer approximation likely to be an equally good approximation for all values of α ?

B.6a 2007

2. A two-dimensional jet of viscous fluid of known momentum flux issues in the x -direction through a narrow nozzle at the origin into an infinite ocean of the same fluid which is at rest at infinity. Starting from the non-dimensional Navier-Stokes equations for steady flow

$$(\mathbf{u} \cdot \nabla) \mathbf{u} = -\nabla p + \frac{1}{Re} \nabla^2 \mathbf{u}, \quad \nabla \cdot \mathbf{u} = 0,$$

where $\mathbf{u} = (u(x, y), v(x, y))$, and assuming that the boundary layer approximation will be valid in the jet sufficiently far downstream, derive the boundary layer equations

$$u \frac{\partial u}{\partial x} + V \frac{\partial u}{\partial Y} = \frac{\partial^2 u}{\partial Y^2}, \quad \frac{\partial u}{\partial x} + \frac{\partial V}{\partial Y} = 0,$$

where V and Y are scaled variables which should be defined.

What are the appropriate boundary conditions at $Y = 0$ and as $Y \rightarrow \infty$?

Show that $\int_{-\infty}^{\infty} u^2 dY$ is constant and explain the physical significance of this result.

Show that there is a similarity solution of the form

$$u = x^\alpha f'(Y/x^\beta),$$

where $\alpha = -\frac{1}{3}$, $\beta = \frac{2}{3}$ and f satisfies the third order ordinary differential equation

$$3f''' + ff'' + (f')^2 = 0,$$

with $f(0) = f''(0) = 0$ and $f'(\infty) = 0$. Hence show that

$$u = 6A^2 x^{-1/3} \operatorname{sech}^2(AYx^{-2/3}),$$

and indicate how the constant A can be determined.

B.6a 2008.

2. (i) The non-dimensional form of the Navier-Stokes equation for steady flow is

$$(\mathbf{u} \cdot \nabla) \mathbf{u} = -\nabla p + \frac{1}{Re} \nabla^2 \mathbf{u},$$

$$\nabla \cdot \mathbf{u} = 0.$$

When the Reynolds number Re is large, derive Prandtl's boundary layer equations for a flow $\mathbf{u} = U(x)\mathbf{i}$ past a flat plate at $y = 0$, $x > 0$, in the form

$$u \frac{\partial u}{\partial x} + V \frac{\partial u}{\partial Y} = U \frac{dU}{dx} + \frac{\partial^2 u}{\partial Y^2}, \quad \frac{\partial u}{\partial x} + \frac{\partial V}{\partial Y} = 0,$$

where the variables Y , V should be defined. State the boundary conditions on u , V . Comment briefly on what happens near $(0, 0)$.

- (ii) When $U(x) = x^m$, show that there is a similarity solution of these boundary layer equations for the stream function ψ of the form

$$\psi = x^\beta f(Y/x^\gamma),$$

where β , γ should be determined, and f satisfies

$$f''' + \beta f f'' + m(1 - f'^2) = 0.$$

What are the boundary conditions for f ?

By defining $w = f'$, find a second order differential equation for $w(f)$. What boundary conditions does $w(f)$ satisfy?

B6.a. 2009

2. The dimensionless two-dimensional steady Navier–Stokes equations are given by

$$\nabla \cdot \mathbf{u} = 0, \quad (\mathbf{u} \cdot \nabla)\mathbf{u} = -\nabla p + \frac{1}{Re} \nabla^2 \mathbf{u},$$

where $\mathbf{u}(x, y, t) = u\mathbf{i} + v\mathbf{j}$ is the velocity, $p(x, y, t)$ is the pressure and Re is the Reynolds number. A uniform stream of fluid with unit upstream velocity in the x -direction flows past a semi-infinite flat plate at $y = 0$, $x > 0$.

(a) When the Reynolds number Re is large, show that at leading order the flow in a boundary layer of thickness of $O(Re^{-1/2})$ on the top of the plate is governed by Prandtl's boundary layer equations

$$\frac{\partial u}{\partial x} + \frac{\partial V}{\partial Y} = 0, \quad u \frac{\partial u}{\partial x} + V \frac{\partial u}{\partial Y} = -\frac{\partial p}{\partial x} + \frac{\partial^2 u}{\partial Y^2}, \quad 0 = -\frac{\partial p}{\partial Y},$$

where the variables Y and V should be defined.

State the boundary conditions on u and V , and deduce that $\partial p / \partial x = 0$.

(b) Denote by $\Psi(x, Y)$ the stream function that satisfies

$$u = \frac{\partial \Psi}{\partial Y}, \quad V = -\frac{\partial \Psi}{\partial x},$$

and assume that Ψ is zero on the plate and strictly increasing with $Y > 0$. By changing dependent variables from (x, Y) to (x, Ψ) , writing $u(x, Y) = U(x, \Psi)$ and using the chain rule, show that

$$\frac{\partial u}{\partial x} = \frac{\partial U}{\partial x} - V \frac{\partial U}{\partial \Psi},$$

and obtain similar expressions for $\partial u / \partial Y$ and $\partial^2 u / \partial Y^2$. Hence show that the boundary layer equations imply

$$\frac{\partial U}{\partial x} = \frac{\partial}{\partial \Psi} \left(U \frac{\partial U}{\partial \Psi} \right).$$

Show that a solution exists of the form $U = g(\eta)$, $\eta = \Psi / \sqrt{x}$, where

$$\frac{d}{d\eta} \left(g \frac{dg}{d\eta} \right) + \frac{1}{2} \eta \frac{dg}{d\eta} = 0.$$

What boundary conditions does $g(\eta)$ satisfy?